

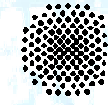
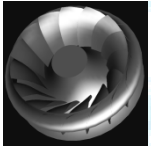
# Parallel in Time simulation of the Navier-Stokes equations using the Finite Element Method

Dipl.-Ing. Konstantinos Ioakimidis

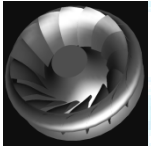
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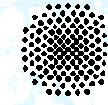
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- Geometry  
problem cases
- Unsteadiness in hydraulic machinery (HM)  
phenomena  
FENFLOSS and some results
- The actual problem
- FENFLOSS  
General information  
Code structure
- New code & algorithm
- Parallel in Time (not working yet)
- Future Work



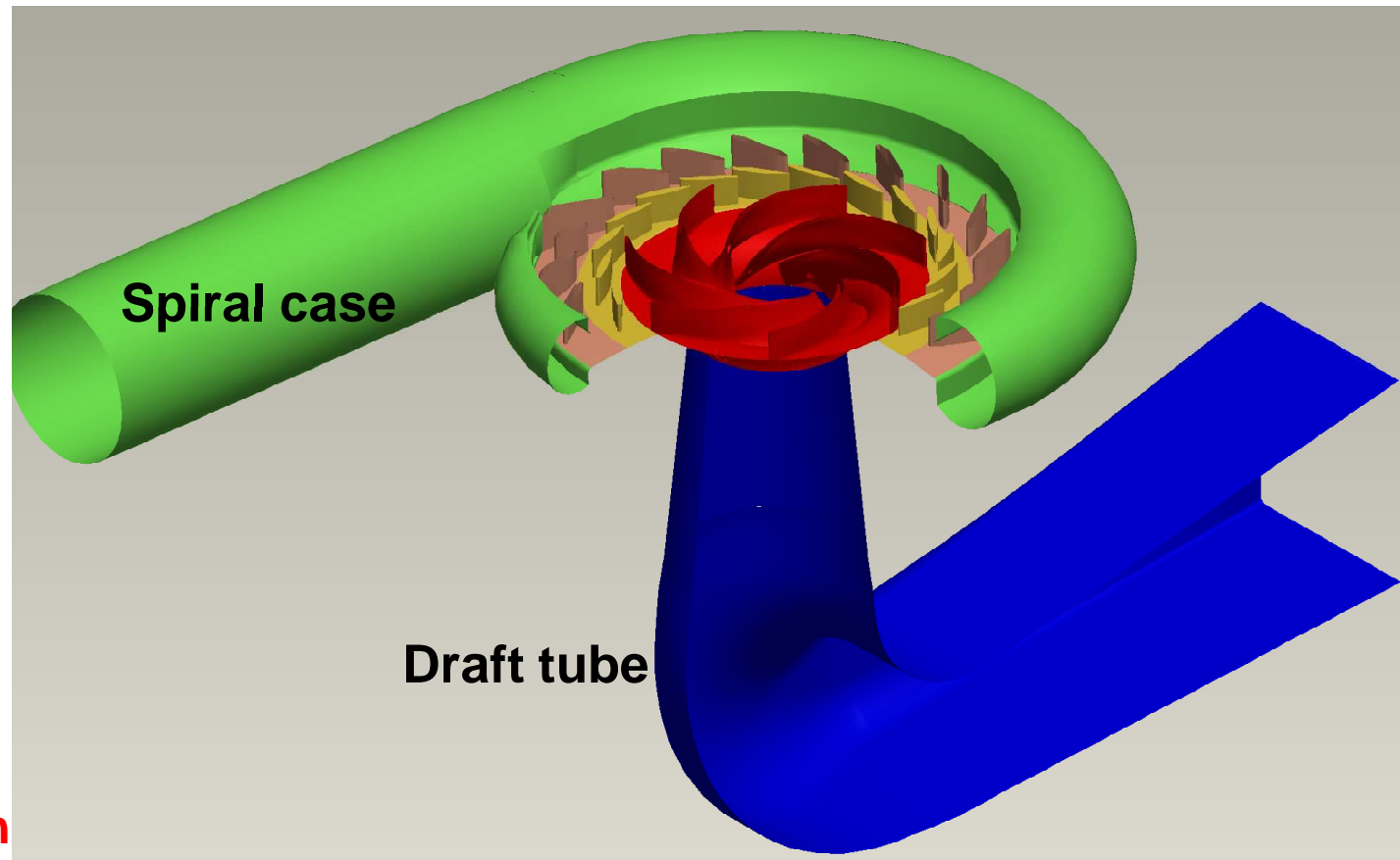
# Geometry



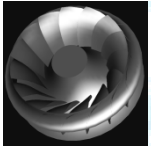
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- Problem size
  - more than 100 million nodes
  - up to 600 million of unknowns for single phase flow

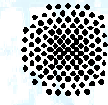
- pipe,
- diffuser



Wall friction  
Swirling flow  
Flow separation



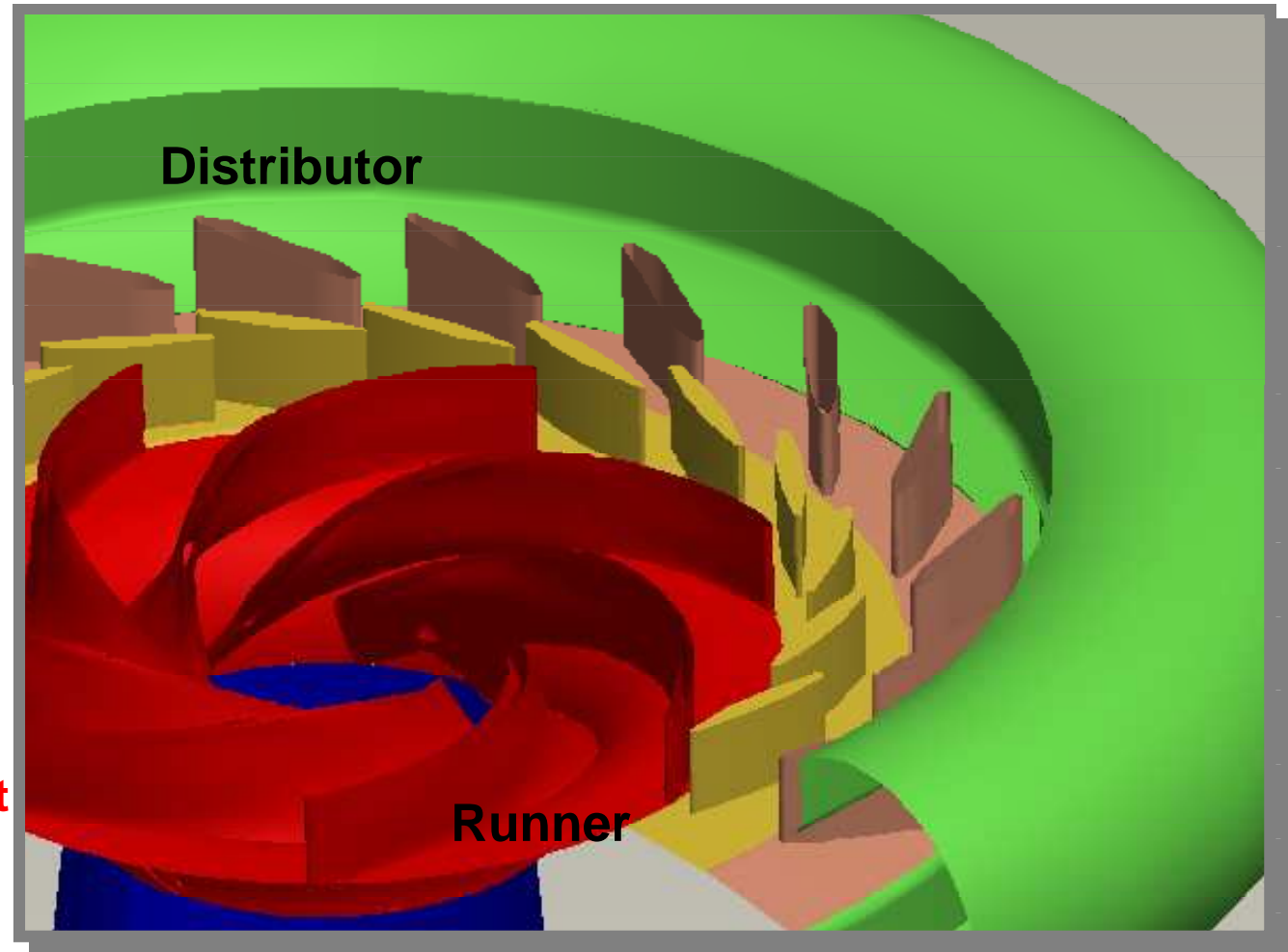
# Geometry zoom in

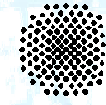
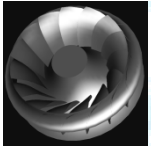


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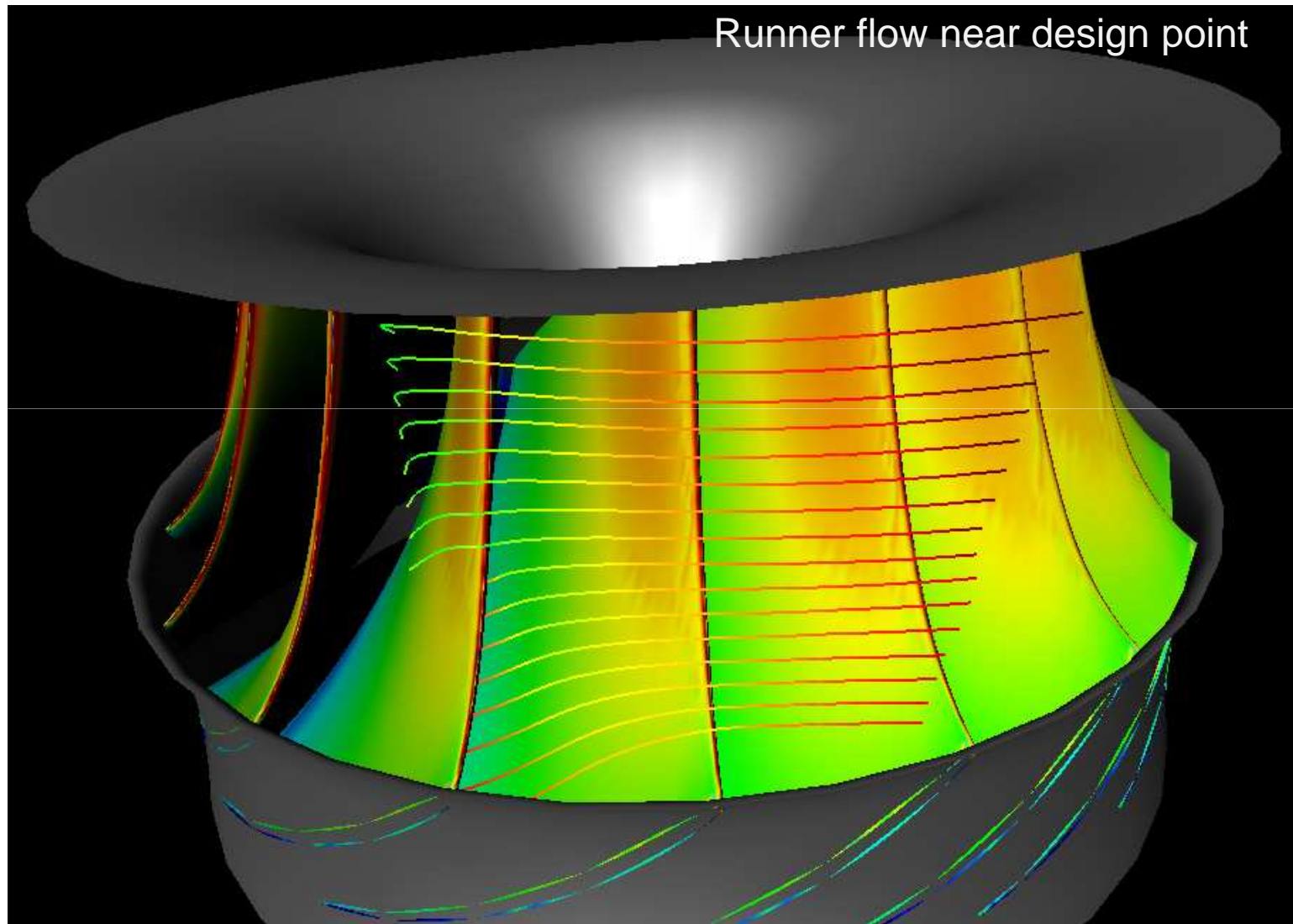
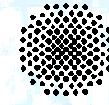
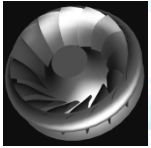
- Hydrofoil
- Rotating Runner
- Gap Flow

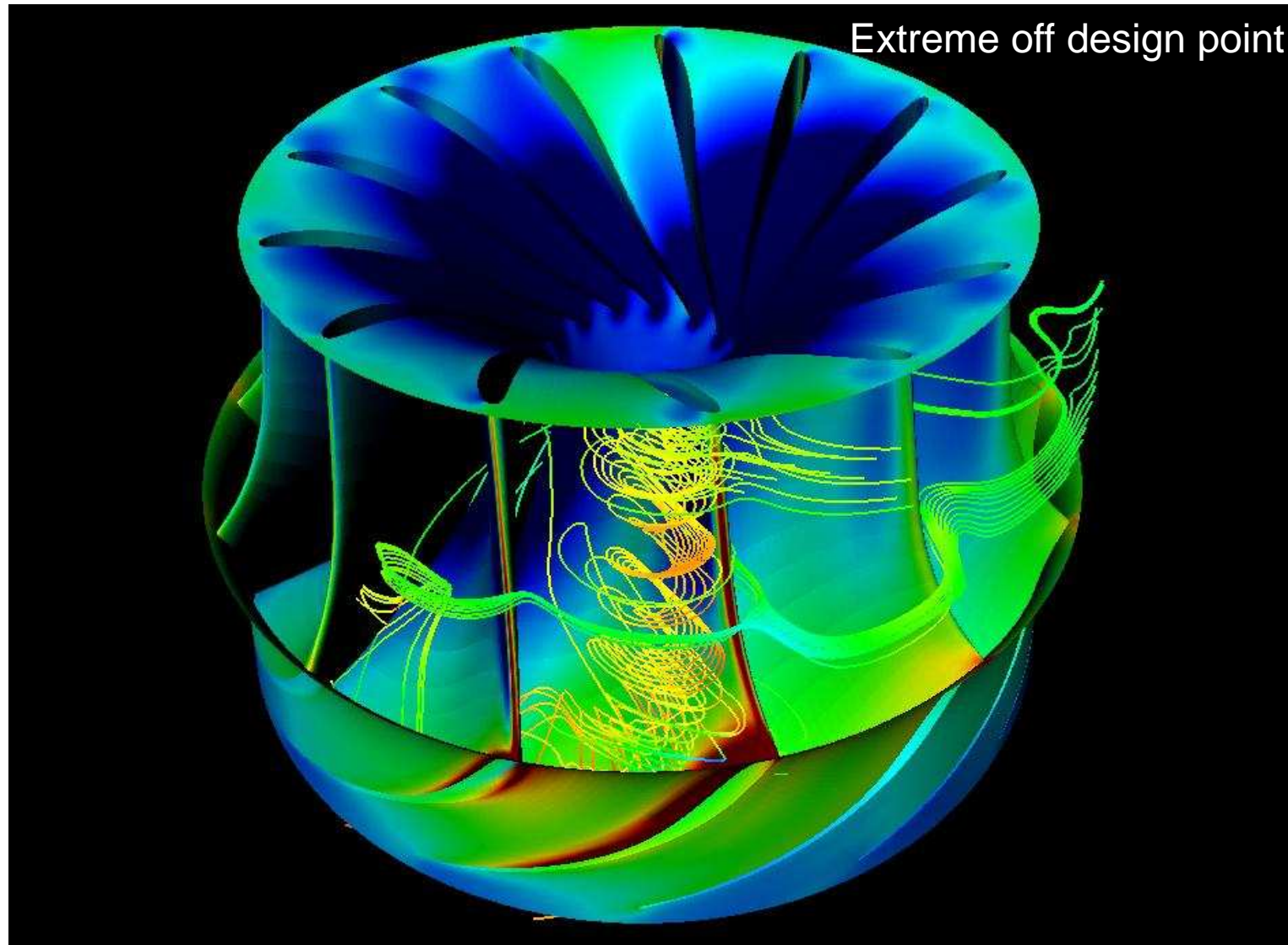
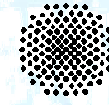
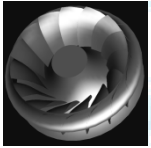
Flow Curvature  
Flow Separation  
Turbulence Transport  
Wake Flow  
Vortex Flow

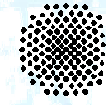
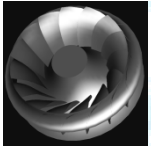




- Unsteady phenomena in hydraulic machinery
  - **Externally forced unsteadiness**  
„forced“ changing geometry  
e. g. Rotor-Stator Interaction, valve closure
  - **Free unsteadiness**  
unsteady vortex movement, flow instabilities  
e. g. draft tube vortex rope, vortex shedding  
e. g. Karman vortex street, flow induced vibrations  
changing geometry caused by unsteady flow.



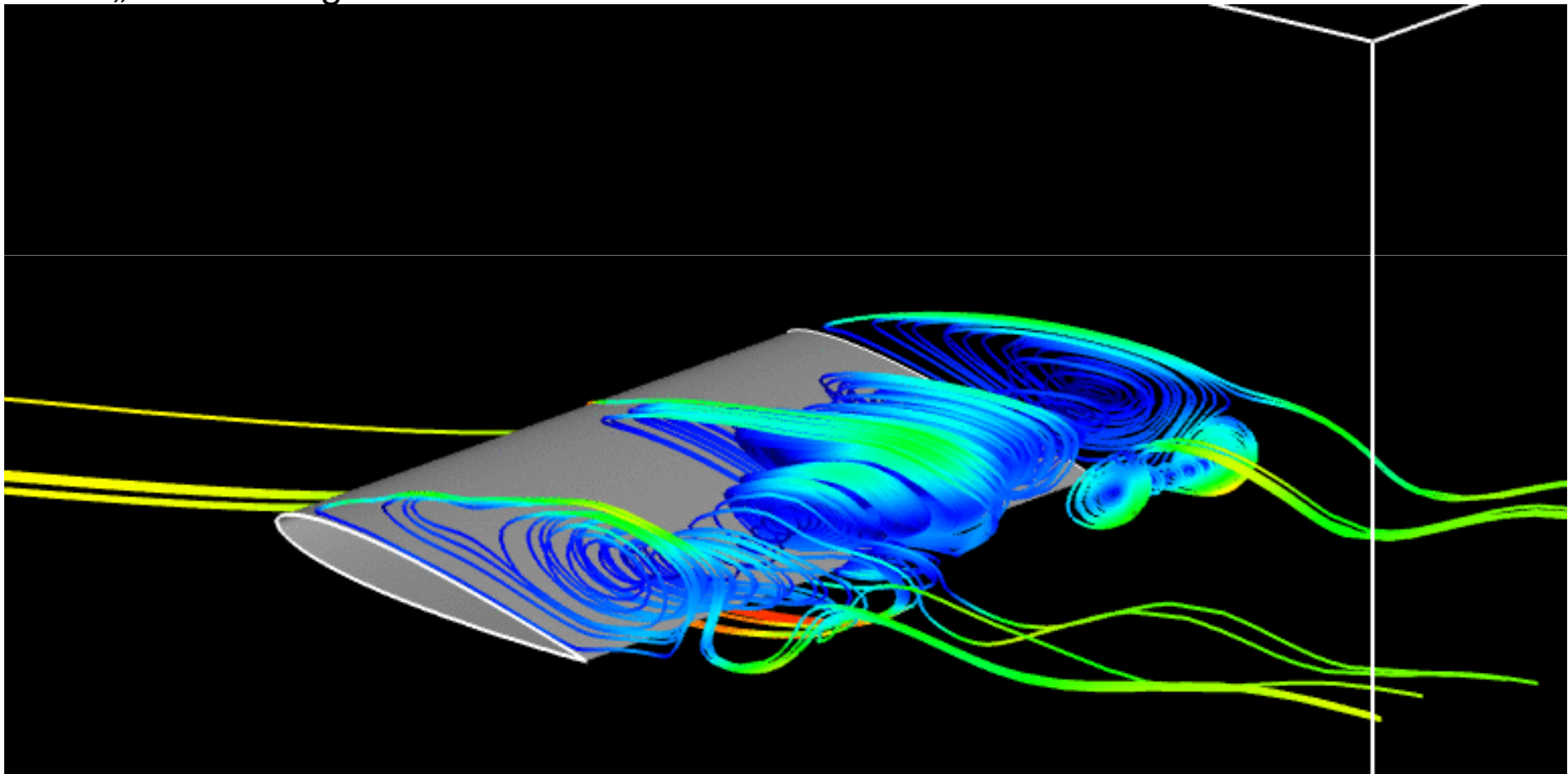


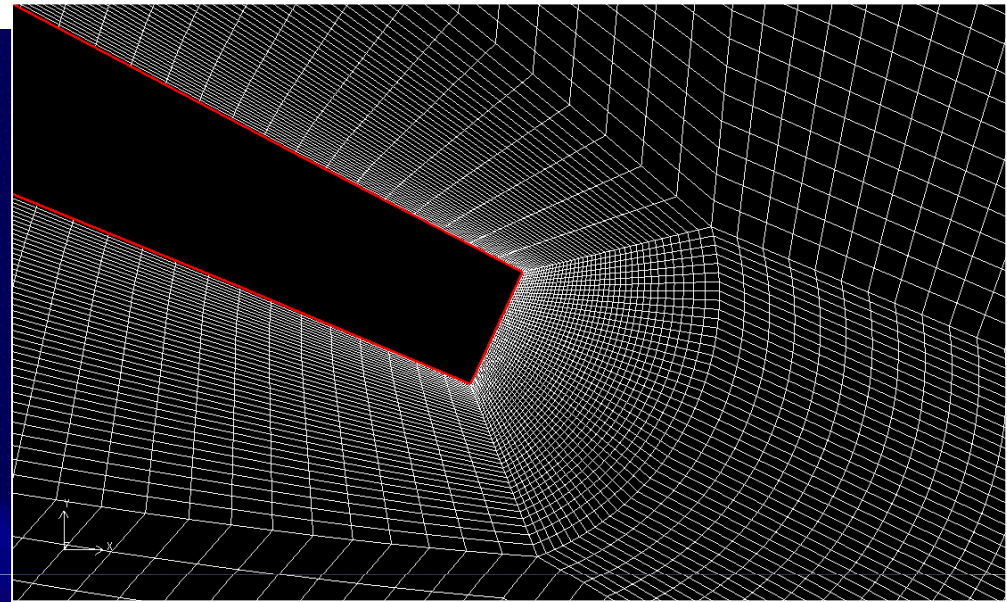
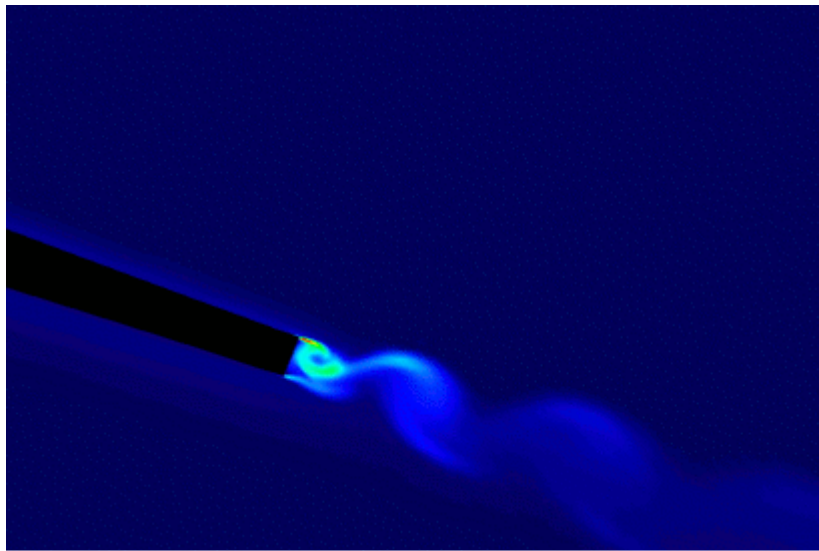
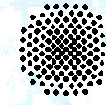
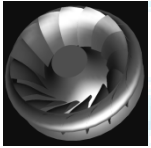


Free unsteadiness

Too high turbulence viscosity will result in steady state recirculation

No „bad“ averaged results

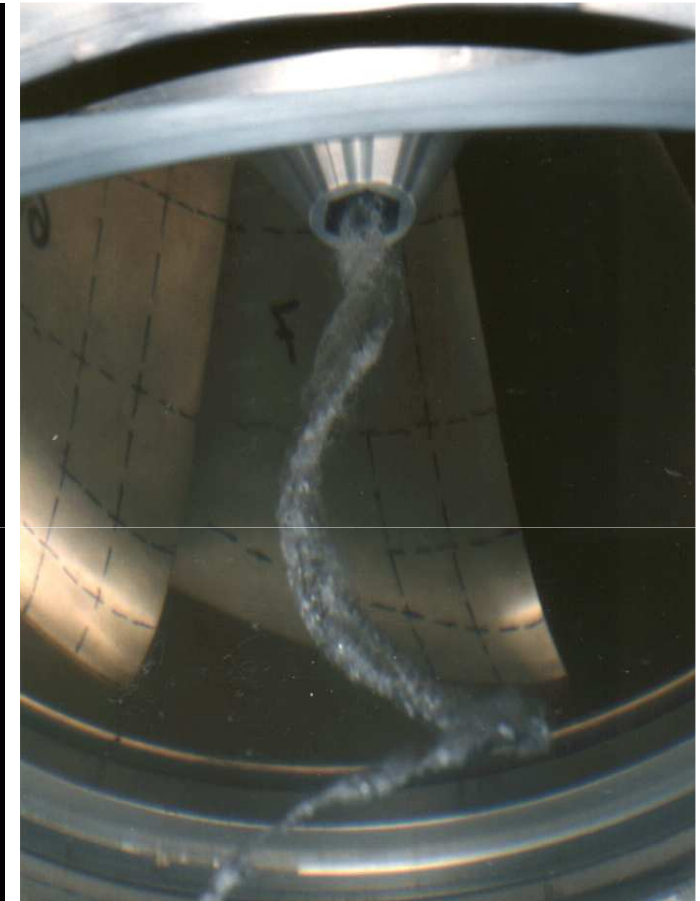
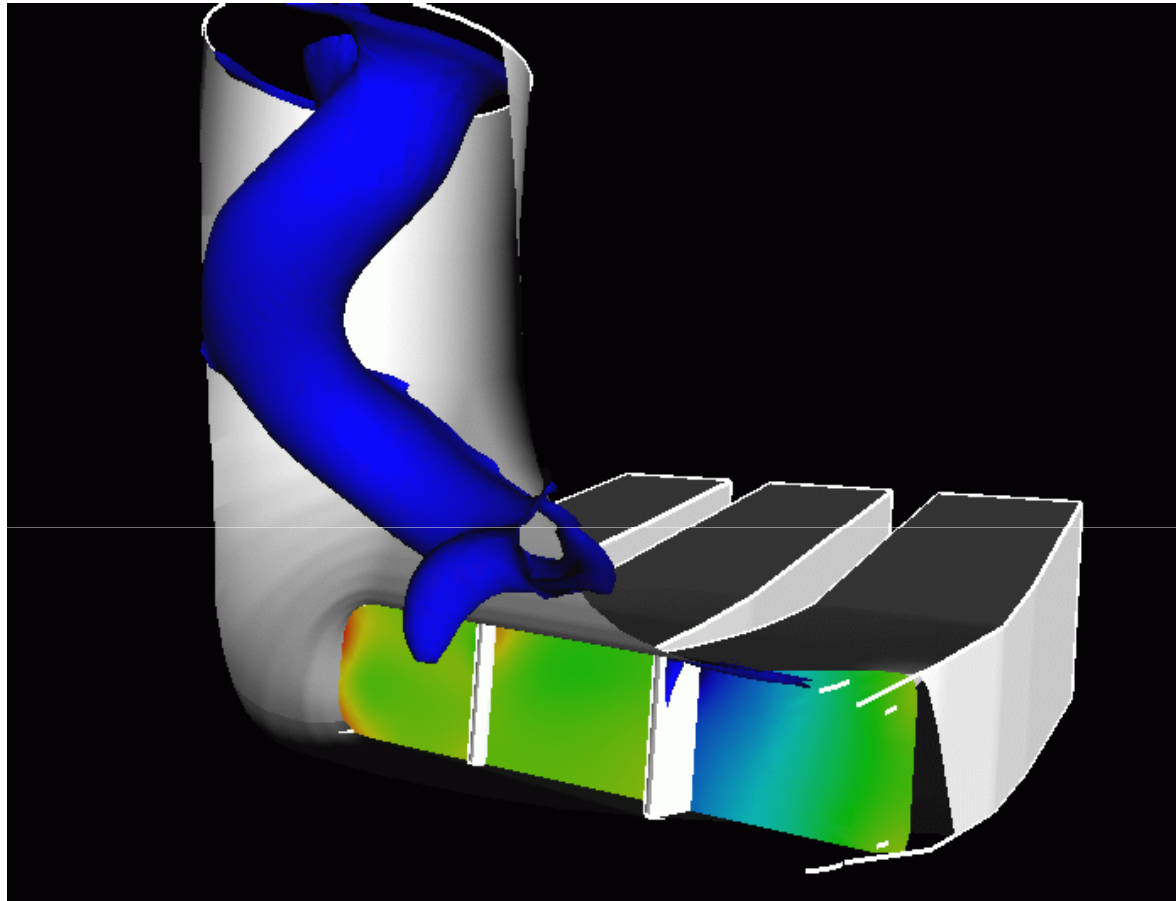
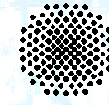
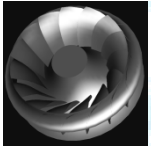




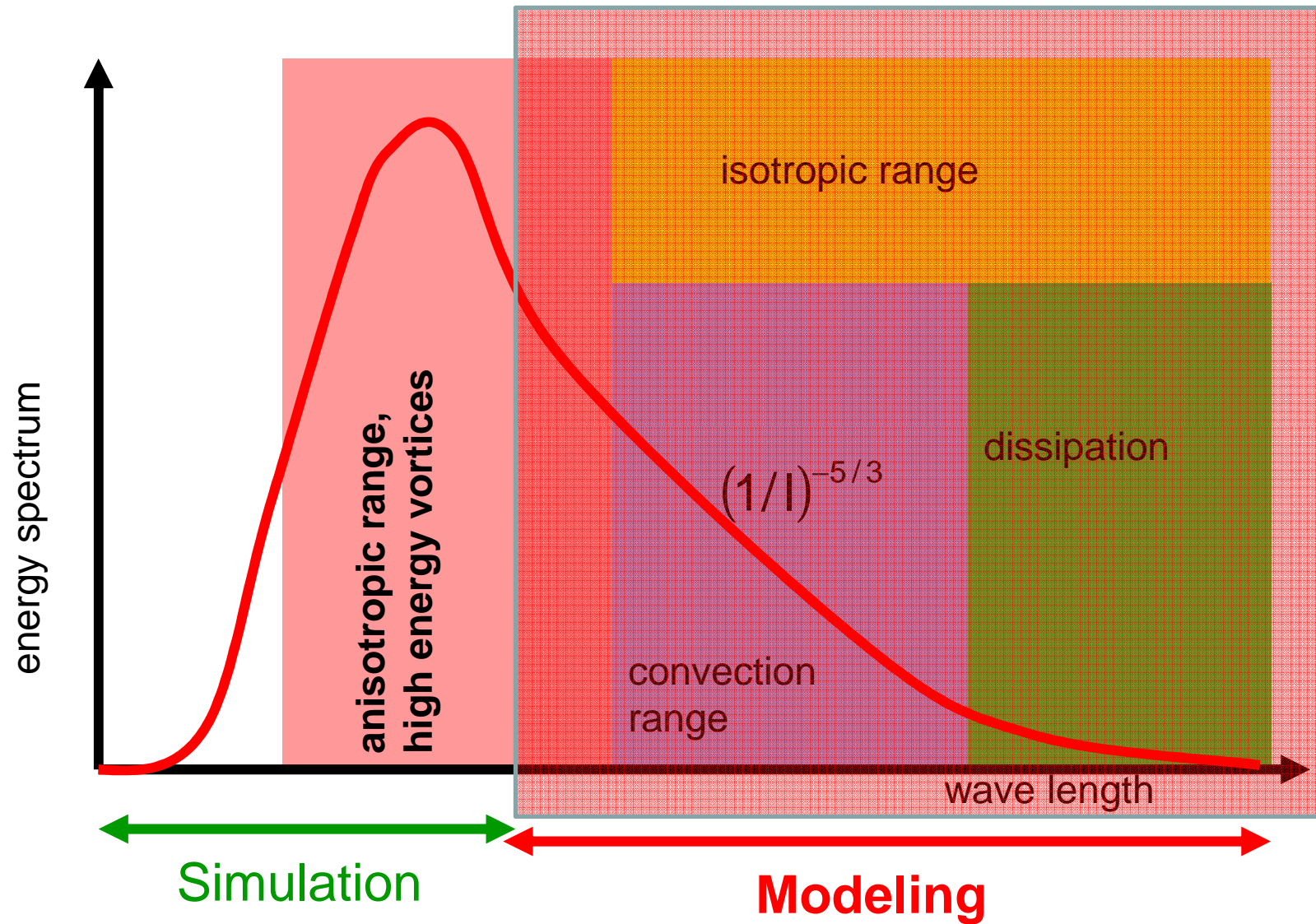
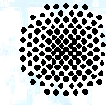
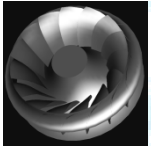
Vortex shedding frequency corresponds with estimation

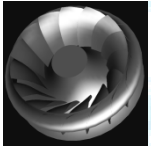
However very fine grids required  
(~ 8 mio nodes)

Too high turbulent viscosity will suppress all unsteadiness => completely wrong results

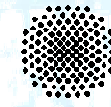


Unstable vortex movement, vortex instability, breakdown  
strong swirling flow, strongly anisotropic turbulence behavior  
Sophisticated turbulence modeling

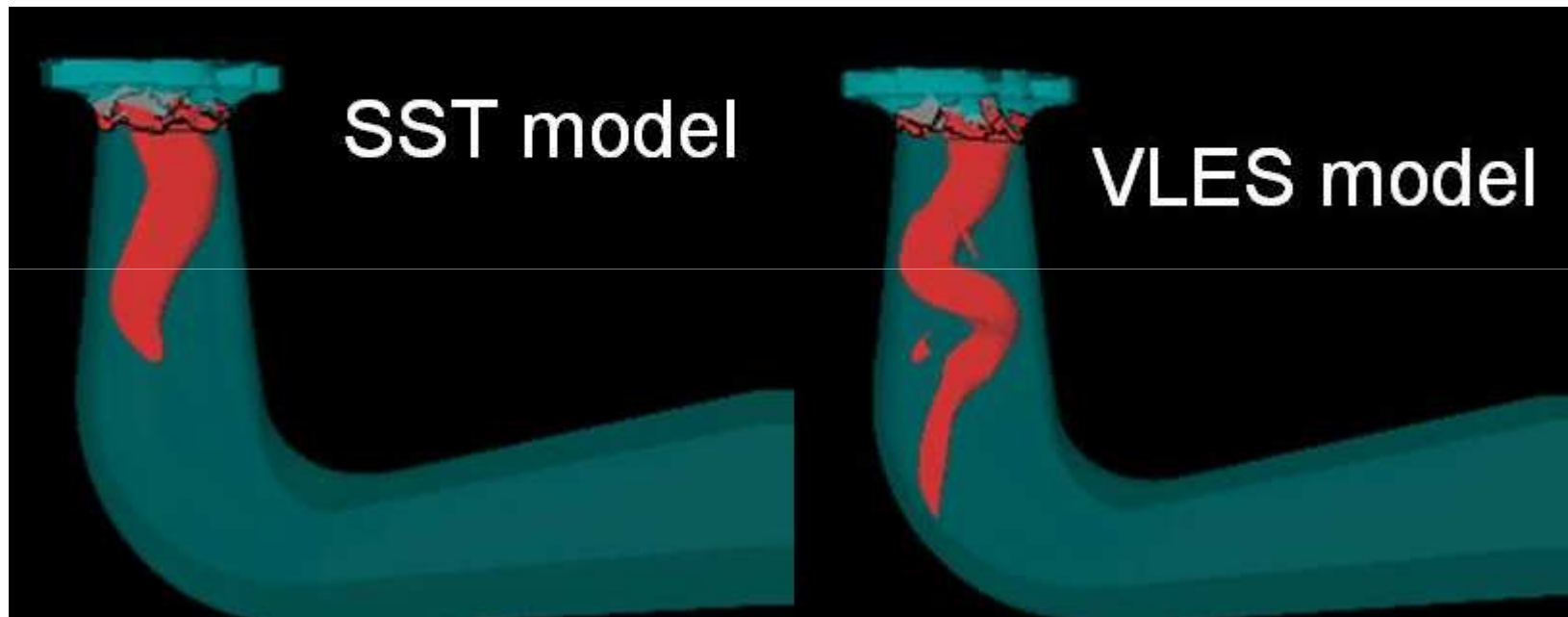




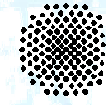
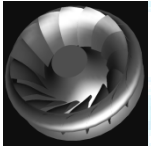
# Comparison Turbulence Models



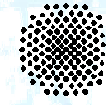
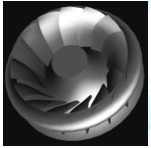
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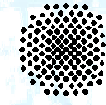
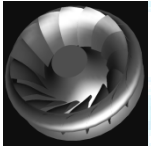
Picture: Huiming Wang, Institute of Fluid Mechanics and Hydraulic Machinery (IHS), Universität Stuttgart



- **Developed at IHS since early 80s**  
Originally designed by Dr. A. Ruprecht still active at IHS  
Parallelised and Vectorised by Dr. M. Maihofer
- **Incompressible Navier Stokes (RANS) Equations**  
Steady and Unsteady Flows  
Laminar as well as Turbulent Flows  
Turbulence Models already implemented  
Simulates any kind of Incompressible flow, especially in Hydraulic Machinery
- **Finite Element Method on Unstructured Meshes**  
a Q1P0 Petrov-Galerkin formulation on hexahedral elements is applied  
pressure correction scheme
- **Can handle Static and Moving Meshes**  
Efficient algorithm implemented for coupling Static & Moving Parts  
e.g. Rotor-Stator coupling
- **Can be coupled with structural mechanics code to Simulate Fluid Structure Interaction (FSI)**  
FSI as well as API interface implemented by Dr. F. Lippold

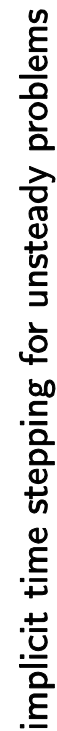


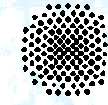
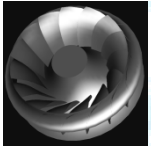
- **Dynamically Load Libraries – API Interface**  
Enables communication with visualisation tools e.g. COVISE (HLRS)  
Exchanging data during the simulation run, in order to automatic visualisation of the time-step results.  
Plenty of Turbomachinery designs made in IHS.
- **The Code is not Publicly available**



FENFLOSS basic building blocks are:

- Read Input (already distributed) files
  - Distributed Memory Approach (DMP) based on MPI
  - Each process reads only its own data from those files
  - Domain Decomposition based on the METIS-library (guarantee good balancing)
  - Advantage -> independence of solution from the number of partitions (speed-up)
- Initialise Data
- Timestep Loop for Unsteady Flows
- Equations System Solving Iterations (Linearisation)
  - Navier – Stokes Equations are non-linear
  - A Richardson Iteration is used to solve the the global non-linear problem
- Assembly of the Global FEM Matrix
  - Time expensive step
- Solving Linear Equation System
  - The linear system is solved by the solver based on Krylov Method (e.g. Van der Vorst's BiCGStab(2))
  - The parallelised & vectorised linear solver handles computations with some millions nodes.

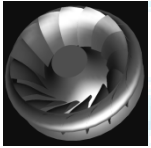




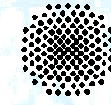
- Performance on the NEC SX-8  
special storage scheme (JAD)  
optimised loops  
-> very efficient on high performance vector computers
- Cooperation with NEC-HLRS TERAFL0P Project  
vector performance of almost 35% of the peak performance  
i.e. 16 GFlop/s per vector CPU

Large Nr. of Processors -> higher communication effort  
decreasing vector length

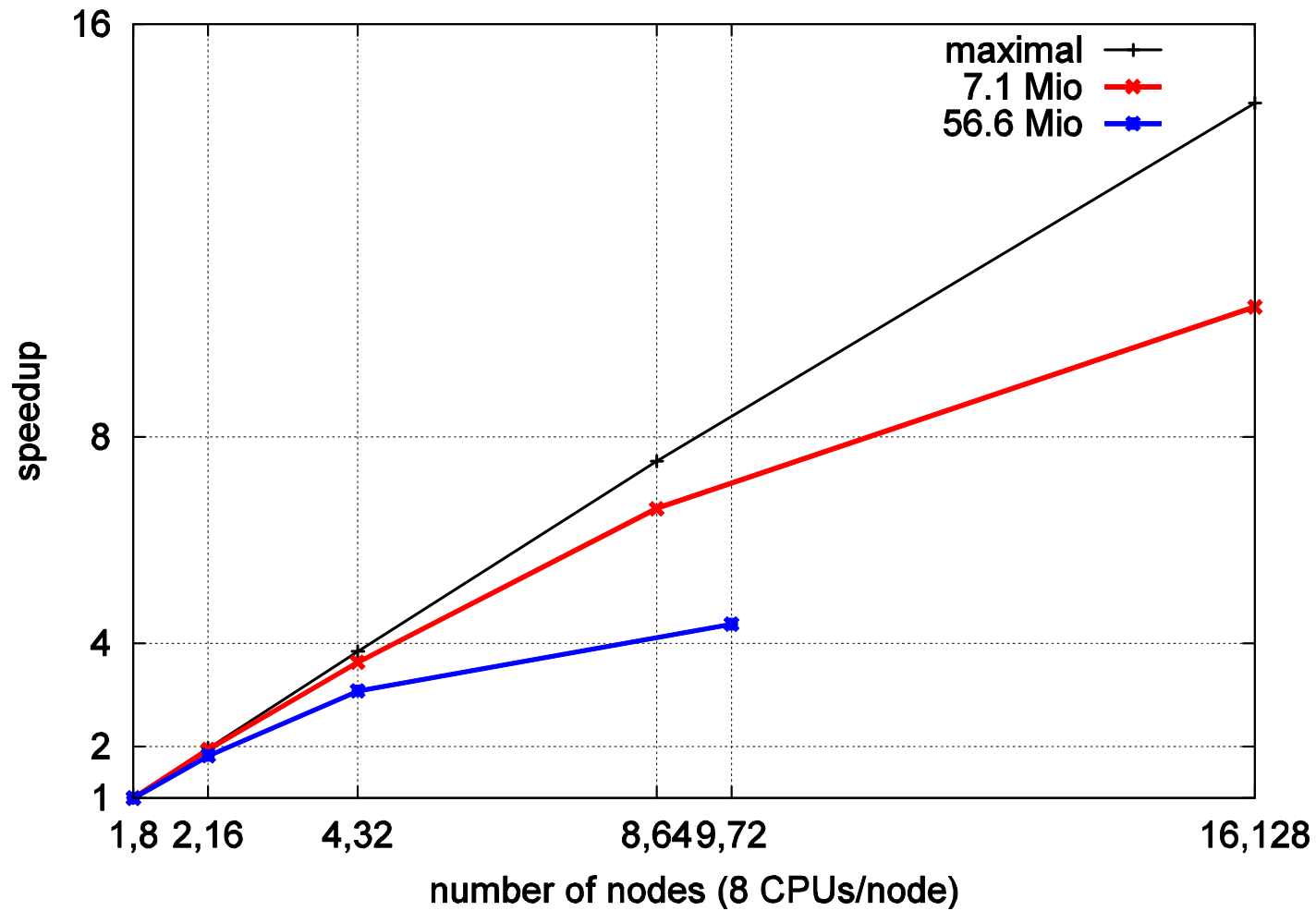
...lower performance

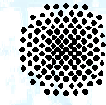
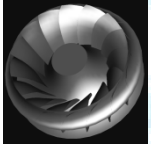


# Speedup II



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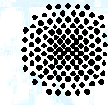
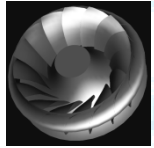




## We need

- To deal with spatial oscillations produced by the discretization of the convection transport terms
- To simplify momentum equation (no pressure gradients)
  - advection-convection problem
- Pressure stability (to use arbitrary interpolation functions) to satisfy the well-known Babuska-Brezzi condition
- Satisfy the continuity
- Implementation simplicity
  - useful for testing algorithms, turbulence models ....
- „Easy“ parallelizable in Space and Time

that is the Characteristic Galerkin Method of Characteristic Based Split



(Greek: ὕδραμα) => rich in water

*call preprocessing*

*DO timestepping*

*call calculate\_timestep/local\_timestep*

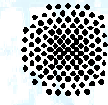
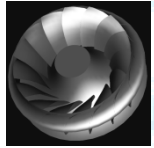
*call momentum\_for\_velocity*

*call poison\_for\_pressure*

*call momentum\_for\_velocity\_correction*

*END DO*

*call postprocessing*



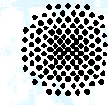
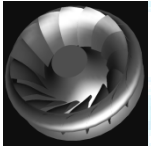
(Greek: ὕδραμα) => rich in water

- Continuity

$$\frac{\partial u_1}{\partial x_1} + \frac{\partial u_2}{\partial x_2} = 0$$

- x Momentum 
$$\begin{aligned} \frac{\partial u_1}{\partial t} + u_1 \frac{\partial u_1}{\partial x_1} + u_2 \frac{\partial u_1}{\partial x_2} \\ = -\frac{1}{\rho} \frac{\partial p}{\partial x_1} + \nu \left( \frac{\partial^2 u_1}{\partial x_1^2} + \frac{\partial^2 u_1}{\partial x_2^2} \right) \end{aligned}$$

- y Momentum 
$$\begin{aligned} \frac{\partial u_2}{\partial t} + u_1 \frac{\partial u_2}{\partial x_1} + u_2 \frac{\partial u_2}{\partial x_2} \\ = -\frac{1}{\rho} \frac{\partial p}{\partial x_2} + \nu \left( \frac{\partial^2 u_2}{\partial x_1^2} + \frac{\partial^2 u_2}{\partial x_2^2} \right) \end{aligned}$$

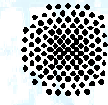
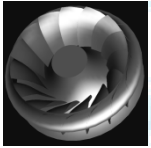


equations of the previous section. If the CG procedure is applied to the above equations, a semi-discrete form of the equations is obtained, namely,

*intermediate  $x_1$  momentum equation*

$$\begin{aligned} \frac{\tilde{u}_1 - u_1^n}{\Delta t} = & -u_1 \frac{\partial u_1^n}{\partial x_1} - u_2 \frac{\partial u_1^n}{\partial x_2} + \nu \left( \frac{\partial^2 u_1}{\partial x_1^2} + \frac{\partial^2 u_1}{\partial x_2^2} \right)^n \\ & + u_1 \frac{\Delta t}{2} \frac{\partial}{\partial x_1} \left[ u_1 \frac{\partial u_1^n}{\partial x_1} + u_2 \frac{\partial u_1^n}{\partial x_2} \right] \\ & + u_2 \frac{\Delta t}{2} \frac{\partial}{\partial x_2} \left[ u_1 \frac{\partial u_1^n}{\partial x_1} + u_2 \frac{\partial u_1^n}{\partial x_2} \right] \end{aligned} \quad (7.129)$$

Roland W. Lewis, Perumal Nithiarasu, Kankanhalli N. Seetharamu, *Fundamentals of the Finite Element Method for Heat and Fluid Flow*, John Wiley & Sons Ltd, 2004



that is, (neglecting third-order terms)

$$\frac{\partial u_1^{n+1}}{\partial x_1} + \frac{\partial u_2^{n+1}}{\partial x_2} - \frac{\partial \tilde{u}_1}{\partial x_1} - \frac{\partial \tilde{u}_2}{\partial x_2} = -\frac{\Delta t}{\rho} \left( \frac{\partial^2 p}{\partial x_1^2} + \frac{\partial^2 p}{\partial x_2^2} \right)^n \quad (7.134)$$

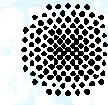
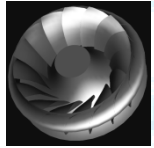
Note that from the continuity equation

$$\frac{\partial u_1^{n+1}}{\partial x_1} + \frac{\partial u_2^{n+1}}{\partial x_2} = 0 \quad (7.135)$$

On substituting the above equation into Equation 7.134, we obtain the pressure equation as follows:

$$\frac{1}{\rho} \left( \frac{\partial^2 p}{\partial x_1^2} + \frac{\partial^2 p}{\partial x_2^2} \right)^n = \frac{1}{\Delta t} \left( \frac{\partial \tilde{u}_1}{\partial x_1} + \frac{\partial \tilde{u}_2}{\partial x_2} \right) \quad (7.136)$$

Roland W. Lewis, Perumal Nithiarasu, Kankanhalli N. Seetharamu, *Fundamentals of the Finite Element Method for Heat and Fluid Flow*, John Wiley & Sons Ltd, 2004



Step 1:

$$[\mathbf{M}] \frac{\Delta \{\tilde{u}_1\}}{\Delta t} = -[\mathbf{C}]\{u_1\}^n - [\mathbf{K}_m]\{u_1\}^n - [\mathbf{K}_s]\{u_1\}^n + \{f_1\}$$

$$[\mathbf{M}] \frac{\Delta \{\tilde{u}_2\}}{\Delta t} = -[\mathbf{C}]\{u_2\}^n - [\mathbf{K}_m]\{u_2\}^n - [\mathbf{K}_s]\{u_2\}^n + \{f_2\}$$

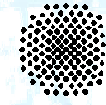
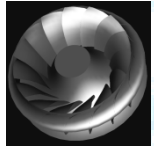
Step 2:

$$[\mathbf{K}]\{p\}^n = -\frac{1}{\Delta t} \left[ [\mathbf{G}_1]\{\tilde{u}_1\} + [\mathbf{G}_2]\{\tilde{u}_2\} \right] + \{f_3\}$$

Step 3:

$$[\mathbf{M}]\{u_1\}^{n+1} = [\mathbf{M}]\{\tilde{u}_1\}^n - \Delta t [\mathbf{G}_1]\{p\}^n$$

$$[\mathbf{M}]\{u_2\}^{n+1} = [\mathbf{M}]\{\tilde{u}_2\}^n - \Delta t [\mathbf{G}_2]\{p\}^n$$



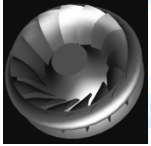
## Semi-explicit Parareal method based on convergence acceleration technique

Loïc MICHEL

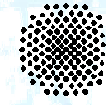
February 18, 2014

### Abstract

The Parareal algorithm is used to solve time-dependent problems considering multiple solvers that may work in parallel. The key feature is a initial rough approximation of the solution that is iteratively refined by the parallel solvers. We report a derivation of the Parareal method that uses a convergence acceleration technique to improve the accuracy of the solution. Our approach uses firstly an explicit ODE solver to perform the parallel computations with different time-steps and then, a decomposition of the solution into specific convergent series, based on an extrapolation method, allows to refine the precision of the solution. Our proposed method exploits basic explicit integration methods, such as for example the explicit Euler scheme, in order to preserve the simplicity of the global parallel algorithm. The first part of the paper outlines the proposed method applied to the simple explicit Euler scheme and then the derivation of the classical Parareal algorithm is discussed and illustrated with numerical examples.



# Conclusions & Future Work



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- Complex phenomena
- We are not Parallel in Time yet!  
as fast as possible
- pfasst and/or parareal
- Performance & Scalability on Cray XE6 (HLRS)
- Fast & Better Results

...Thank you for your attention ! ! !